

NASA Institute for Advanced Concepts

Large Telescope Using a Holographically-Corrected Membrane Mirror

(LTHM)

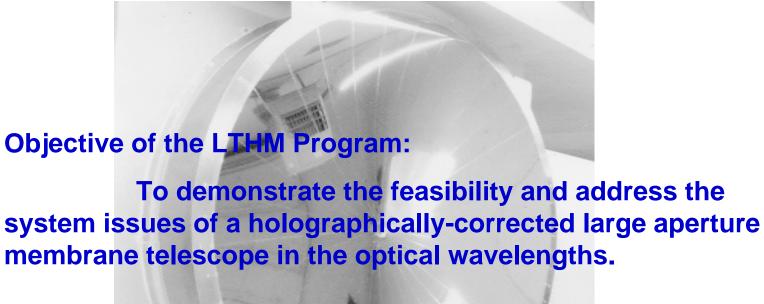
Arthur L. Palisoc

L'Garde, Inc.

Tustin, California

June 6 - 7, 2000









Inflatable Antenna Experiment





Membrane Mirrors



3 m diameter HAIR reflector (on-axis)



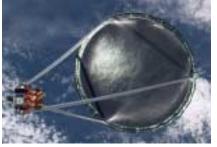
1/5th sector of the IAE reflector (off-axis)

LTHM

Large Telescope Using Holographically-Corrected Membrane Mirror State of the Art Surface Precision



Inflatable Antenna Experiment (IAE)



D=50 ft (offset), F/D=1 As-built rms: 1.5 mm RMS: center 11 meters

9.8 ft IAE Sector



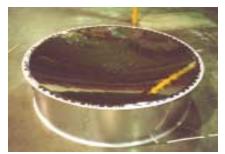
D=9.8 ft, F/D=1 As-built rms: 0.6 mm RMS

IRD Inflatable Reflector



D=23 ft, F/D=1/2 As-built rms: 1.2 mm RMS

LIS Inflatable Reflector



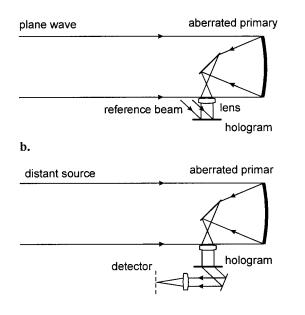
D=9.8 ft, F/D=1/2 As-built rms: 0.86 mm RMS

LTHM



Holographic Correction





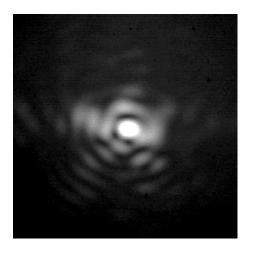
(a) Recording: collimated light illuminates the aberrated primary to form the object beam. The hologram is written with a reference beam incident at an angle.

(b) Reconstruction: starlight (distant object) produces a reconstructed beam, which is focused to produce an unaberrated image.



Holographic Correction

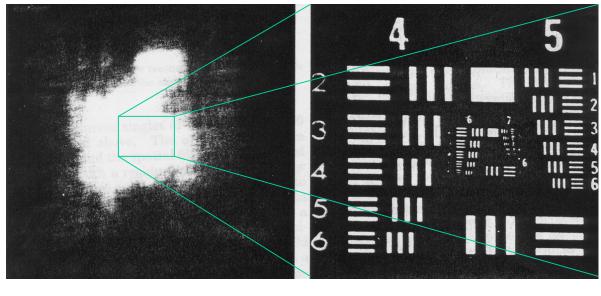




(a) (b) Focal spot images: (*a*) Before correction (actual size). (*b*) After correction (magnified 450X).



Holographic Correction



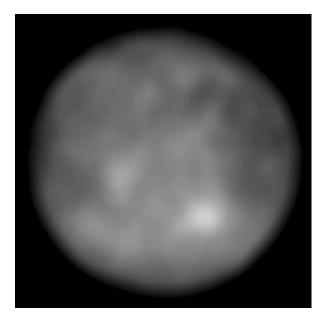
(a)

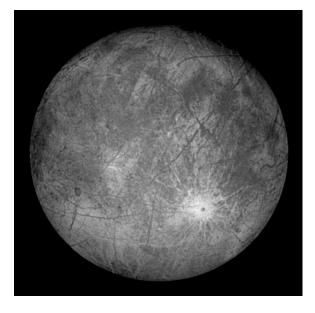
(b)

1951 USAF resolution chart before and after holographic correction.



Simulation using a 100m diameter Holographically-Corrected Telescope





Europa as viewed by the Hubble Space Telescope

Europa as viewed by a 100 m Holographically-Corrected Telescope



Test Facilities

- L'Garde, Inc., Tustin, California
 - Analysis, design, & fabrication of membrane mirror
 - Membrane materials testing
 - Surface profile measurement of membrane mirror

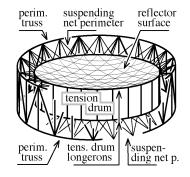
USAF Academy, Colorado Springs

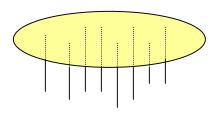
Holographic tests and correction of membrane mirror

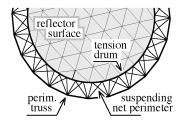


Membrane Mirror Configurations





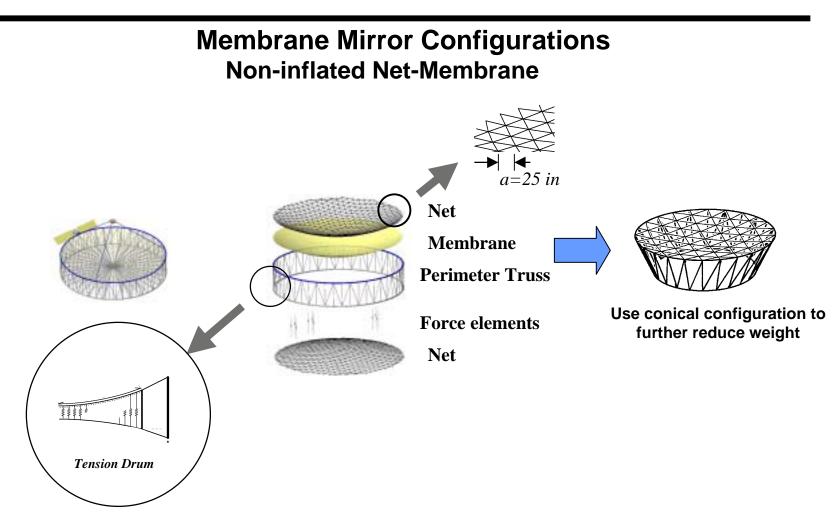




Inflatable Net-Membrane

Non-inflated Net-Membrane Configuration Non-inflated Net-less Configuration

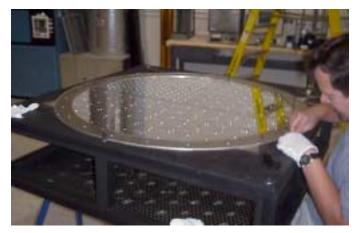




L'Garde, Inc. Proprietary



The Net-Less Membrane Mirror







L'Garde, Inc. Proprietary



Membrane Mirror Configurations

Purely Inflatable

- Smoothest surface
- Highest surface accuracy
- Canopy obscures signal
- Needs makeup gas
- Simplest to manufacture
- Packageable into the smallest volume

Net-Membrane

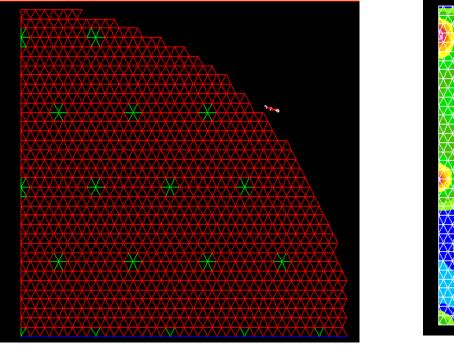
- Nearly flat triangular facets
- Moderate surface accuracy
- No canopy needed
- No makeup gas needed
- More labor intensive
- Packageable into a small volume.

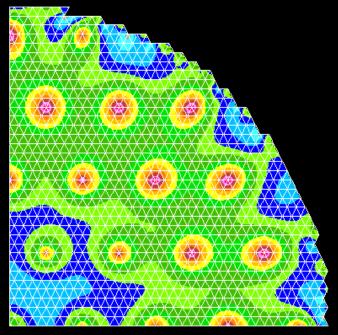
Net-Less Membrane

- Cusps at tab locations
- Moderate surface accuracy
- No canopy needed
- No makeup gas needed
- Simpler than netmembrane
- Packaging volume is smaller than that of Netmembrane but larger than that of purely inflatable



Finite Element Simulation of Net-Less Membrane Mirror





Predicted Surface Accuracy: $\varepsilon = 0.25 \text{ mm RMS}$



Phase I Tasks

- Conceptual design of a compact, space-based membrane telescope that incorporates a real-time holographic correction.
- Build on the work for the NRO where we will prove the holographic correction of a 1 m diameter membrane telescope.
- Analytically characterize the net-membrane and net-less membrane concepts sensitivity analyses.
- Investigate the production of holograms in real-time and at several wavelengths *in-situ*; e.g. use of photopolymers.
- Compare performance relative to each other: *inflatable* v.s. *net-membrane* v.s. *net-less membrane* configuration.
- Identify and address the system issues.
- Chart a roadmap to an orbiting 10 m diameter imaging telescope using holographically-corrected membrane mirrors.



System Issues

- Real-Time Holographic Correction
- Bandwidth
- Wideband holographic correction
- Holographic materials
 - > photopolymers
 - > FBAG
 - > OASLM
- Laser beacon source "fixed" versus trailing
- Single hologram recorded at multiple wavelengths
- CTE and creep of membrane material PBO has extremely low CTE.
- Space environment resistance



Phase II Plans

• Build a 1m diameter membrane telescope with holographic correction *in-situ*.

• Carry out a full static and dynamic analysis of the concept selected for a 10 m diameter.

• Continue to investigate *Real-Time* holographic materials – suitability of photopolymers as a holographic medium in a space environment.

• Investigate the possibility of using a distant laser source in space – "trailing" or fixed at the *ISS* for example.

- Feasibility of a simple, secondary adaptive optics system.
- Modularizing and shielding of the telescope for optimum performance.