Acknowledgements

Artificial Neural Membrane Flapping Wing
Phase I Study
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Los Alamos National Laboratory Technology Transfer Office
Artificial Neural Membrane
Flapping Wing: Introduction

Highly complex, non-linear concept.

Integrates (seemingly) unrelated technologies.

Requires new methods in mathematical biology and new applications in applied general topology.

Is highly dependent on emerging technologies.
Artificial Neural Membrane Flapping Wing: Introduction

Proposed to investigate the development of an artificial neural membrane (ANM) flapping wing as a demonstration of the potential of neural engineering and advanced materials to create a revolutionary technology based in biomimetics and nanoscience.
Artificial Neural Membrane Flapping Wing: Introduction

Artificial Neural Membrane Flapping Wing (ANMF)
New class of intelligent functional structures

Experience in functional structures
Current resident experience and technology

Identification of primary thrust areas for development
Materials & Structures, Computational & Simulation, Aerodynamics

Identification resources supporting thrust areas
Establish current and emerging science including resources to for further study

Artificial Neural Membrane (ANM) Technology
Applications and potential
Proposal conceptualized configuration of carbon nanotubules embedded in the outer polymer capsule of a membrane matrix.

Integration concept based on ARSI’s force measurement and A-neuron sensors (concept right).

Simple bi-model or quasi-steady state flapping wing.

Initial concept of a thin film layered matrix with embedded nems.
Advances in materials and computational methods have provided a new environment for the development of artificial neurons as device resident controllers for structures and vehicles.
The ARSI Systems Engineering Laboratory with developed A-neuron sensors for optronic and detector systems.
Proposal was based on the success artificial neuron or A-neuron sensor concept developed by ARSI to provide a stable, thin film detector for IRST and coherent light hazards.
In the 1980’s developed the Neurogenesis Algorithms from the study of the differentiation of chick embryos and the development of the CNS and PNS of vertebrates.

Neurogenesis models were developed from observational data of developmental anatomical changes of embryos as well as signal analysis of neurons in vivo and in vitro.
Experience in Multifunctional Smart Structures

Extensive experience in engineering thin film, optical and composite sensors. Including:

- High-temperature
- Variable frequency
- Embedded sensors
- Radiation detectors
- Board level process monitoring

ARSI μ-G Sensor
0.2 mm

ARSI Physiological Sensor
Named Sophia meaning “knowledge” and incorporating the ARSI Neurogenesis Module that integrates self-generating neural networks.

Based on topological algorithms that allow for the simulation of a network on a network.

Current Phase I study investigated network morphing and simulation using topological and algebraic mapping methods.
Applications

Near Term Applications (5-10 years)
- Global Environmental Monitoring System
- ARSI Sophia GEMS Concept

Future Applications (10-20 years)
- ARSI Martian MAV Planetary Remote Sensing & Communication Systems
- Orbital ANM Ion Propelled Vehicle
The artificial neural membrane or ANM flapping wing is seen as a multifunctional communications and remote sensing platform for planetary exploration.
Artificial Neural Membrane
Flapping Wing: Concept

Robust multifunctional platforms covering hundreds or thousand of square kilometers providing a Delay Tolerant Network (DTN) for planetary studies and support for communications, weather, radiation monitoring for future astronauts.
Artificial Neural Membrane
Flapping Wing: Concept

Martian Aircraft

1970's design was generated by NASA Dryden based on a hydrazine-fueled version of the Mini-Sniffer aircraft.

In the early 1990's, a solar aircraft concept was generated by NASA Glenn.
Future Martian Microaerial Vehicles (MAV) will need to be fault tolerant, provide robust communications & remote sensing information and cover large areas.
Advantages of airborne platforms for planetary studies:

- Airborne platforms can achieve science objectives difficult to complete from orbit or from surface rovers.
- They can cover much larger distances in a single mission and are not limited by the terrain.
- Improved imaging and sensor missions using airborne platforms.
- Produce images of more than a magnitude higher resolution than state-of-the-art orbiting spacecraft.
Artificial Neural Membrane
Flapping Wing: Concept

Revolutionary neural systems combining biomimetics and nanoscience.

Adaptive A-Neural Systems provide an advanced learning platforms for investigating unknown environments.
Artificial Neural Membrane
Flapping Wing: Concept

Martian weather and atmospheric sampling.

Radiation survey of Mars.

Near IR spectrometry, essential to analyzing mineral deposits.

High spatial resolution magnetometry, that may provides answers to the origin of high crustal magnetism seen from orbit.
Thrust areas are specifically directed at developing complex functional nanostructures, a new class of intelligent structure.

- Materials & Structures
- Computational & Simulation
- Aerodynamics
**ANM Flapping Wing: Thrust Areas**

**Materials & Structures**
Mechanics, force distribution, interaction of nanoscale structures, integration of nems

**Computational Aspects**
Governing equation for a quasi-steady state flapping wing, neural algorithms, topological networks, control systems

**Aerodynamics**
Airfoil, stability, flow distribution, performance in earth and Martian atmospheres
Wing Geometry & Aerodynamics: Computational & Physical Aspects

Chose a flapping wing insect model to provide a consistent proven computational model with physical data for ANM Flapping Wing Study.

In order to determine the potential for an ARSI Neural Controller to recognize flapping wing flight with a given geometry.
Quasi-Steady State Flapping Wing Model

Chose because of consistently valid research supporting the model.

Based on physical models fabricated and tested by Sanjay Sane and Michael Dickinson, University of California-Berkeley.

Provided rigorous and well defined computational models.

This wing geometry lends itself to higher stability for a membrane based vehicles.
“2-Dimensional motion in an inviscid fluid, the first term was calculated for each blade element and integrated along the span of the wing” thus providing an estimate 3-D forces on the airfoil (Sane & Dickinson, 2002).

\[ F_a = \rho \frac{\pi}{4} R^2 \hat{c}^2 (\hat{\phi} \sin \alpha + \hat{\phi} \alpha \cos \alpha) \int_0^1 \hat{r} \hat{\omega}^2(\hat{r}) d\hat{r} - \alpha \rho \frac{\pi}{16} \hat{c}^3 R \int_0^1 \hat{\omega}^2(\hat{r}) d\hat{r} \]

Quasi-steady transitional estimate for the net force was developed through vector addition of lift & drag estimates (Sane & Dickinson, 2002).
Completed three experiments to evaluate the capability of the Neurogenesis Algorithms to recognize flapping wing motion. 2- and 3-Dimensionsional algorithms were used.
Wing Geometry & Aerodynamics: Computational & Physical Aspects
NiTi

Nitinol (NiTi-55% nickel + 45% titanium) belongs to a class of materials called Shaped memory Alloys (SMA).

Shaped Memory Effect (SME) allows the alloy to remember a particular shape. Once a shape has been remembered, the alloy may be bent out of shape. Then returned to its original shape by heating the alloy to its transition temperature.
NiTi Physical Properties:

- Tensile Strength --- 200,000 psi
- Melting Point --- 1,250°C (2,282°F)
- Resistance --- 1.25 ohms per inch (.006 wire)
- Corrosion Resistant
The NiTi alloy has nearly equal amounts of both metals. However changing the ratio will significantly effect on the transition temperature.

A 1-% difference in the ratio varies the transition temperature from -100 to +100 C.

NiTi wire in our experiment has a transition temperature of 70C.
The generated reverting force for one square inch of NiTi is +30,000 PSI.

NiTi-Flexinol® wire (Dynalloy Inc.) is used as an actuator material. Flexinol wire can last 1,000,000 cycles whereas other NiTi alloy wire that may last 1,000 cycles or less. 15-mil Flexinol wire has a rated contractive force of 1.786 kg (4 lbs.)
Wing Geometry & Aerodynamics

Major Issues:

Flutter and Vibration.

Dynamic stability.

Adequate control feedback.
Los Alamos National Laboratory
Center for Integrated Nanotechnologies (CINT)

Definition:

Complex and collective interactions between individual components in materials to yield emergent properties
Complex Functional Nanostructures

- Complex Functional Nanomaterials
  - Interactions between individual components, properties and functions.

- Nanomechanics
  - Mechanical behavior of nanoscale materials and structures

- Computation & Simulation
  - Computational methods and advanced computer simulation to develop and define a new class of function structures.
Complex Functional Nanostructures

Force interactions on the wing structure by the mechanically functional components.

Embedded carbon nanotubules or deposited NiTi structures.

Substrate interactions and fatigue.

Delamination of thin film stacks.
Example of Enabling Technologies

IR photovoltaic nanotechnology from the University of Toronto, 4nm cells produce nearly the same efficiency per area as UV solar technology.
Concept of the IR solar technology coated with a selective coating varying UV & IR radiation according to component application in the substrate.

Graphic by ARSI simulating IR photovoltaic system with proprietary frequency and wavelength selective coating.
ARSI Sophia Technology
Applications

- Dropsonde Sophia Cloud Rider System
- A-Neural Membrane Ion Thrusters
- And future applications.
Near Term Applications (5-10 years)

Global Environmental Monitoring System (GEMS) Concept

ARSI Sophia Cloud Rider GEMS Concept
- Dropsonde system long-duration weather monitoring.
- Proliferant emissions detection, homeland security and defense missions.
- Geo-Environmental and atmospheric health monitoring.

Covers large area through dispersion canister. Uses the physics of clouds to “hitch” a ride.
Current US space suits do not provide protection against electrical shock, radiation or crush injuries.

ANM functional structures using deposited alloys and/or carbon nanotubules will provide greater protection and act as an assistive system with adaptive geometry actuators, etc.
Remote physiological and environmental monitoring of astronauts.

Some Applications:

- Telemedicine reporting system.
- Improved countermeasures effectiveness.
- Radiation dosimetry.
Future systems:

**Advanced A-neural functional structures:**

Artificial organs, board level and membrane enzyme and protein production, membrane satellites and mirrors, and space habitats.
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